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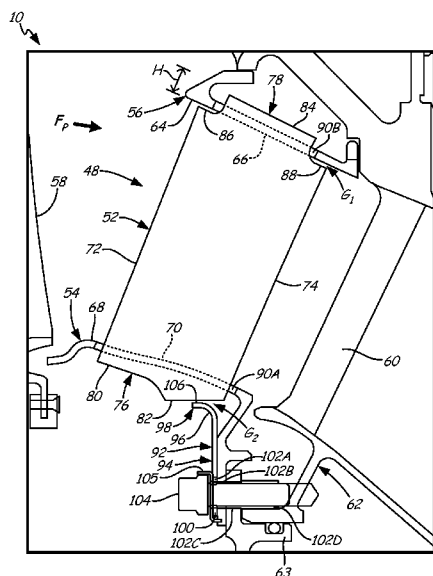


Fig. 2

(57) Abstract: An assembly for use in a gas turbine engine includes an inner diameter (ID) shroud having a plurality of ID slots and an outer diameter (OD) shroud having a plurality of OD slots and positioned radially outward from the ID shroud. A plurality of stator airfoils extend from the ID shroud to the OD shroud. Each of the stator airfoils is positioned at least partially in one of the ID slots and is positioned at least partially in one of the OD slots. A bumper is positioned proximate a first end of each of the stator airfoils so as to limit movement of the stator airfoils in a radial direction.