

HIGH-ENTROPY OXIDES FOR THERMAL BARRIER COATING (TBC) TOP COATS

CROSS-REFERENCE TO RELATED APPLICATIONS

[0001] The present application claims the benefit of U.S. Provisional Patent Application No. 62/743,392 filed October 9, 2018, the disclosure of which is expressly incorporated by reference herein in its entirety.

TECHNICAL FIELD

[0002] The present invention relates to top coat materials for thermal barrier coatings (TBCs) with excellent phase stability and dimensional stability over a wide temperature range, and good thermal cycling performance with low heat conductivity. The top coat materials may be in the form of a powder, alloy, top coat or coating, and may be employed in a thermal spray powder with bond coating materials to obtain a thermal barrier coating (TBC) system. The present invention also relates to a method for reducing delamination of a top coat from a bond coat and substrate, such as a gas turbine engine component.

BACKGROUND OF THE INVENTION

[0003] A complete thermal barrier coating system includes a top coat, such as a thermal barrier coating (TBC), and a bond coat or bond layer. A common bond coat is made of MCrAlY alloy, where M represents Ni, Co, Fe or their combinations. To increase the performance of a bond coat, Hf, Re and Pt and various other rare earth elements are often added in advanced bond coats. A common top coat is made of zirconia (ZrO_2) stabilized by one or more of yttrium oxide, ytterbium oxide, cerium oxide, titanium oxide, magnesium oxide, calcium oxide, lanthanum oxide, or dysprosium oxide, or of gadolinium zirconate ($\text{Gd}_2\text{Zr}_2\text{O}_7$).

[0004] The TBC system is applied and bonded to a substrate, such as a superalloy and protects the substrate in a hot and harsh environment such as in a gas turbine engine environment. The bond coat or bond layer is between the top coat and the substrate, and bonds the top coat to the substrate. The bond coat or bond layer is formed from a bond coating material, which may be in powder form for application to the substrate. The bond coat or bond layer formed from the bond coating material affects the thermal cyclic fatigue and sulfidation resistance of the top coat, such as a TBC, which may be evaluated for

effectiveness by the furnace cyclic lifetime of the TBC with and without the presence of sulfur. A TBC can be degraded due to high temperatures and a harsh environment, such as the presence of sulfur. For example, using oil containing a high sulfur content as fuel in an industrial gas turbine having a TBC is one important factor reducing TBC lifetime.

[0005] One important failure of a complete TBC system happens at the bond coat/top interface. When a TBC system is exposed to a high temperature, an oxide layer, called a thermally grown oxide (TGO), nucleates and grows between the bond coat and top coat, to block oxygen to further diffuse inward to prevent the substrate from oxidation. A dense alpha-alumina layer of TGO is desired because it effectively blocks oxygen diffusion inward and also grows slowly itself. However, there are significant differences in thermal expansion coefficients of the top coat, bond coat and TGO. Significant internal stresses caused by temperature changes are built up at the top coat/ TGO interface and the TGO/bond coat interface while a TBC system experiences thermal cycles (room temperature to operation temperature to room temperature). As the TGO layer thickens, the thermal internal stresses become higher and higher, and eventually the top coat, such as the TBC, fails due to delamination by thermal internal stresses. To increase performance and energy efficiency, modern jet engines and industrial gas turbines are seeking a higher operation temperature, therefore a higher thermal shock resistance of a TBC system is desired.

[0006] It has been found that changes in crystal structure or a phase change when subjected to temperature changes from room temperature to higher operation temperatures and *vice versa*, for example from a tetragonal crystal structure to a cubic or fluorite crystal structure or phase or *vice versa*, can be accompanied by a substantial change in volume of a top coat which can cause deleterious delamination of the top coat from a bond coat or substrate.

[0007] Thermal barrier coatings (TBCs) are a critical technology for aero engines and industrial gas turbines. The baseline TBC system comprises a (Ni, Co, Fe)CrAlY bond coat with an oxide ceramic top coat. The top coat is typically a Y_2O_3 -stabilized ZrO_2 with 7-8 wt% Y_2O_3 . Key TBC requirements are: low thermal conductivity; phase and dimensional stability across the operating temperature range; resistance to sintering at the high end of the operating temperature range; and good oxidation and thermal cycling performance of the combined coating system. For aero engines resistance to 'CMAS' attack is also important. Molten CMAS (calcia–magnesia–aluminosilica) deposits ingested with intake air penetrate and interact with thermal barrier coatings (TBCs) in gas-turbine engines to degrade the zirconia-

based TBCs and the overall TBC performance. As gas turbine engines are driven to operate at higher temperatures to maximize efficiency, components become more susceptible to attack by deposits of calcium magnesium alumino-silicate (CMAS). Molten CMAS is known to interact with TBCs both thermochemically, by dissolving the ceramic and reprecipitating it as a new or modified phase, and thermomechanically, by infiltrating the porosity and degrading the strain tolerance.

[0008] Many variants of thermal barrier coating (TBC) top coat chemistry described above are in use today. This baseline TBC system is, however, now reaching its limits, particularly in terms of thermal conductivity.

[0009] High entropy oxides (HEOs) are oxides with a high configurational entropy, S_{config} . They typically contain five or more different metal cation types as well as oxygen, to form one or more oxide sublattices. HEOs have high levels of lattice distortion and other lattice defects. This reduces thermal conductivity and can improve mechanical properties, such as toughness. The configurational entropy S_{config} of an HEO is generally $1.5R$ per mole or greater, where R is the gas constant $8.314 \text{ J.K}^{-1}\text{mol}^{-1}$; this definition using the value of S_{config} being a commonly-accepted definition of a high-entropy material.

[0010] Numerous thermal barrier coatings containing multiple metal oxides and lower heat conductivity are known, but their use as a top coat having a single phase which does not undergo a phase change over a wide temperature range is not disclosed. For example, U.S. Patent No. 6,812,176 to Zhu et al, the disclosure of which is herein incorporated by reference in its entirety, discloses a thermal barrier coating composition which is about 46-97 molar percent base oxide, about 2-25 molar percent primary stabilizer, about 0.5-12.5 molar percent group A dopant, and about 0.5-12.5 molar percent group B dopant. The base oxide is selected from the group consisting of ZrO_2 , HfO_2 , and combinations thereof. The primary stabilizer dopant is selected from the group consisting of Y_2O_3 , Dy_2O_3 , and Er_2O_3 and combinations thereof. The group A dopant is selected from the group consisting of alkaline earth oxides, transition metal oxides, rare earth oxides and combinations thereof. The group B dopant is selected from the group consisting of Nd_2O_3 , Sm_2O_3 , Gd_2O_3 , Eu_2O_3 and combinations thereof. The Zhu patent does not disclose the compositions as high entropy oxides (HEOs) or as having an S_{config} greater than $1.5R$.

[0011] U.S. Patent Nos. 7,001,859 and 7,186,466 each to Zhu et al, the disclosures of which are each herein incorporated by reference in their entirety, each disclose thermal barrier

coating compositions having 46–97 molar percent base oxide, 2–25 molar percent primary stabilizer, 0.5–25 molar percent group A dopant, and 0.5–25 molar percent group B dopant. The base oxide is selected from the group consisting of ZrO_2 , HfO_2 and combinations thereof, the primary stabilizer is selected from the group consisting of Y_2O_3 , Dy_2O_3 , Er_2O_3 and combinations thereof; the group B dopant is selected from the group consisting of Nd_2O_3 , Sm_2O_3 , Gd_2O_3 , Eu_2O_3 and combinations thereof; and the group A dopant is selected from the group consisting of rare earth oxides, alkaline earth metal oxides, transition metal oxides and combinations thereof, but excluding those species contained in the base oxide, group B dopant and primary stabilizer groups. The ratio of the molar percentages of group A dopant to group B dopant in the composition is between about 1:10 and about 10:1. Neither of the Zhu patents disclose their compositions as high entropy oxides (HEOs) or as having an S_{config} greater than 1.5R.

[0012] The high entropy (S_{config} is greater than 1.5R) can be calculated for any composition using standard thermodynamic formulae, as described in, for example C. M. Rost, *Ph.D thesis, North Carolina State Univ (2016)*, "Entropically-stabilized oxides: Explorations of a novel class of multicomponent materials," the disclosure of which is herein incorporated by reference in its entirety.

[0013] U.S. Patent No. 7,001,859 to Dorfman et al, and U.S. Patent No. 9,975,812 to Doesburg et al, the disclosures of which are herein incorporated by reference in their entireties, each discloses a ceramic material for use in thermal barriers for high temperature cycling applications and high temperature abradable coatings. The material is an alloy formed predominantly from ultra-pure stabilized zirconia (ZrO_2) and/or hafnia (HfO_2) alloys that have uncharacteristically high sintering resistance to achieve a high service lifetime. It is disclosed that changes in the coating microstructure over the in-service lifetime are retarded. The material has about 4 to 20 weight percent of a stabilizer of one or more rare earth oxides; and a balance of at least one of zirconia (ZrO_2), hafnia (HfO_2) and combinations thereof, wherein the zirconia (ZrO_2) and/or hafnia (HfO_2) is partially stabilized by the stabilizer, and wherein the total amount of impurities is less than or equal to 0.15 weight percent. The patents disclose that zirconia alloys have: 1) some of the highest melting points of all ceramics, and this means theoretically some of the highest temperatures for which the onset of sintering occurs, 2) one of the lowest thermal conductivities of all ceramics, and 3) one of the highest coefficients of thermal expansion of all ceramics, so it is most compatible with transition metal alloys during thermal cycling. However, according to the patents, zirconia alone cannot

fulfill the coating requirements because it undergoes a phase transformation from tetragonal to monoclinic during thermal cycling. This transformation is presumed to cause a detrimental volume change resulting in large strain differences between the coating and the substrate. When the resulting stresses exceed the bond strength of the coating to the substrate, the coating will detach. For this reason a phase stabilizer is added to the zirconia and/or hafnia, such as yttria, which suppresses the tetragonal to monoclinic phase transformation. The compositions are not disclosed as high entropy oxides (HEOs) or as having an S_{config} greater than 1.5R.

[0014] U.S. Patent Application Publication Nos. 2018/0022928 and 2018/0022929 each to Blush, the disclosures of which are herein incorporated by reference in their entireties, disclose coated articles supporting high-entropy nitride and/or oxide thin film inclusive coatings. The high-entropy alloys systems are heat stable and may be used in optical coatings. A first material system that may be used includes SiAlN with one or more (and preferably two or more) of elements such as Hf, Y, Zr, Ti, Ta, and Nb. A second material system that may be used includes TiO, with one or more (and preferably two or more) of elements such as Fe, Co, Ni, Sn, Zn, and N. The material systems may in some cases be high-index materials that can serve as a substitute for titanium oxide in layer stacks. It is disclosed that current high-entropy alloys are known to have high temperature stability because of extremely high entropic contributions. This is related to their equiatomic or near-equiatomic compositions, and the high number of elemental constituents. It is known that $\Delta G = \Delta H - T\Delta S$ (where ΔG is the change in Gibbs free energy, ΔH is enthalpy, T is temperature, and ΔS is entropy). The phase with the lowest Gibbs free energy of formation will be the phase formed at equilibrium, so increasing entropy will increase the likelihood of a phase being stable. According to Blush, generally, traditional low entropy materials have a ΔS_{config} of about 1 R (or sometimes lower), medium entropy materials have a ΔS_{config} of about 1 R to about 1.5 R, and high entropy materials have a ΔS_{config} of greater than about 1.5 R. However, it is disclosed, the lines between low and medium, and medium and high, need not be precisely delineated. For instance, some materials may have four constituent materials nonetheless may be considered high-entropy for these purposes, even though ΔS_{config} might ordinarily be expected to be slightly less than 1.5 R. Use of the compositions in TBC systems is not disclosed.

[0015] U.S. Patent Application Publication No. 2018/0128952 to Yeh, the disclosure of which is herein incorporated by reference in its entirety, discloses a multi-film structure coated on the surface of a workpiece, where the multi-film structure is formed, for example,

by making a high-entropy material film of at least two layers and a non-high-entropy material film of at least one layer stacked on each other. The high-entropy material film may be a high-entropy alloy film, high-entropy nitride film, high-entropy carbide film, high-entropy nitroxide film, high-entropy carbonitride film, high-entropy oxide film, high-entropy carbon oxide film, and other high-entropy ceramic films. Exemplary high-entropy films disclosed are a high-entropy alloy film having the equiatomic composition of AlCrNbSiTi and a thickness of 0.25 μm , a high-entropy nitride film having the composition of (AlCrNbSiTi)N, i.e. (Al₁₀Cr₁₀Nb₁₀Si₁₀Ti₁₀)N₅₀ and a thickness of 0.2 μm , a high-entropy nitride film having the composition of (CrNbSiTiZr)N and a thickness of 0.15 μm , a high-entropy alloy film having the composition of AlCrNbSiTi and a thickness of 0.8 μm , a high-entropy oxide film having the composition of (AlCrNbSiTi)₄₀O₆₀ and a thickness of 0.2 μm , a high-entropy alloy film having the composition of AlCrNbSiTiZr and a thickness of 0.4 μm , a high-entropy carbonitride film having the composition of (AlCrNbSiTiZr)₅₀C₂₀N₃₀ and a thickness of 0.4 μm , a high-entropy carbon oxynitride film having the composition of (AlCrNbSiTiZr)₄₀C₂₀N₃₀O₂₀ and a thickness of 0.6 μm , a high-entropy nitride film having the composition of (AlCrNbSiTiZr)N and a thickness of 0.2 μm , and a high-entropy carbide film having the composition of (CrNbSiTiZr)C and a thickness of 0.2 μm . Use of the compositions in TBC systems is not disclosed.

[0016] Metal oxides of the formula MO, where "M" represents 5 or more oxide-forming metals, which are HEOs having the rock salt "NaCl" crystal lattice structure are disclosed in the following articles, the disclosures of each of which are herein incorporated by reference in their entireties:

1. C. M. Rost, *Ph.D thesis, North Carolina State Univ (2016)*, "Entropically-stabilized oxides: Explorations of a novel class of multicomponent materials."
2. C. M. Rost, E. Sachet, T. Borman, A. Moballeggh, E. Dickey, D. Hou, J. Jones, S. Curtarolo, J. P. Maria, *Nature Communications*: 09-25-2015, "Entropy-stabilized oxides."
3. Moballeggh, C. M. Rost, Jon-Paul Maria, E C. Dickey, *Microsc. Microanal.*, 21 (2015), pp. 1349-1350: "Chemical homogeneity in entropy-stabilized complex metal oxides."
4. Z. Rak, J-P, Maria, D. W. Brenner, *Mater Lett*: 217 (2018) pp. 300-303: "Evidence for Jahn-Teller compression in the (Mg,Co,Ni,Cu,Zn)O entropy."

5. C. M. Rost, Z. Rak, D. W. Brenner J.-P. Maria, *J. Am Ceramic Society*, 100(2017), pp. 2732-2738, "Local structure of the $\text{Mg}_x\text{Ni}_x\text{Co}_x\text{Cu}_x\text{Zn}_x$ ($x=0.2$) entropy-stabilized oxide: An EXAFS study."
6. Z. Rak, C. M. Rost, M. Lim, P. Sarker, C. Toher, S. Curtarolo, J. P. Maria, D. W. Brenner, *J. App. l Phys.*, 120 (2016) pp. 95-105, "Charge compensation and electrostatic transferability in three entropy-stabilized oxides: results from density functional theory calculations."
7. G. Anand, A. P. Wynn, C. M. Handley, C. L. Freeman, *Acta Mater.*, 146(2018) pp. 119-125, "Phase stability and distortion in high entropy oxides."
8. Sarkar, R. Djenadic, N. J. Usharani, K. P. Sanghvi, *J. Euro Ceram Soc*, 37(2017) pp. 747-754, "Nanocrystalline multicomponent entropy stabilized transition metal oxide."
9. D. Berardan, S. Franger, D. Dragoë, A. K. Meena and N. Dragoë, *Phys. Status Solidi RRL* 10, 4(2016), pp. 328-333, "Colossal dielectric constant in high entropy oxides."
10. D. Berardan, S. Franger, A. K. Meena and N. Dragoë, *J. Mater. Chem. A*, 24(2016), pp. 9536-9541, "Room temperature Lithium superionic conductivity in high entropy oxides."
11. D. Berardan, A. K. Meena, S. Franger, C. Herrero and N. Dragoë, *J. Alloys and Compounds*, 704(2017) pp. 693-700, "Controlled Jahn-Teller distortion in $(\text{MgCoNiCuZn})\text{O}$ -based high entropy oxides."
12. Sarkar, L. Velasco, D. Wang, Q. Wang, G. Talasila, L. de Biasi, C. Kubel, T. Brezesinski, S. Bhattacharya, H. Hahn, B. Breitung, *Nature Communications*: 08-24-2018, "High entropy oxides for reversible energy storage."
13. A. Giri, J. Braun, C. M. Rost, P. E Hopkins, *Scripta Mater.*, 138(2017) 134-138, "On the minimum limit to thermal conductivity of multi-atom component crystalline solid solutions based on impurity mass scattering."

Use of the compositions in TBC systems is not disclosed in the articles.

[0017] Metal oxides of the formula MO_2 , where "M" represents 5 or more oxide-forming metals, which are HEOs having the fluorite " CaF_2 " crystal lattice structure are disclosed in the following articles, the disclosures of each of which are herein incorporated herein by reference in their entireties:

14. R. Djenadic, A. Sarkar, O. Clemens, C. Loho, M. Botros, V. Chakravadhanula, C. Kubel, S. Bhattacharya, A. Gandhi, H. Hahn, ***Mater. Res. Lett.*** 5(2017), pp. 102-109, "Multicomponent equiatomic rare earth oxides."
15. K. Chen, X. Pei, L. Tang, H. Cheng, Z. Li, C. Li, X. Zhang, L. An, ***J. Euro Ceram Soc***, 38(2018) pp. 4161-64, "A five-component entropy-stabilized fluorite oxide."
16. A. Sarkar, C. Loho, L., Velasco, T. Thomas; S. Bhattacharya, H. Hahn, R. Djenadic, ***Dalton Transactions*** 36(2017), pp. 12167-176, "Multicomponent equiatomic rare earth oxides."

Use of the compositions in TBC systems is not disclosed in the articles.

[0018] ABO₃ type oxides where A and B are cations, which are HEOs having the perovskite crystal lattice structure are disclosed in the following articles, the disclosures of each of which are incorporated herein by reference in their entireties:

17. S. Jiang, T. Hu, J. Gild, N. Zhou, J. Nie, M. Qin, T. Harrington, K. Vecchio, J. Luo, ***Scripta Mater***, 142(2018), pp. 116-120, "A new class of high-entropy perovskite oxides."
18. A. Sarkar, R. Djenadic, D. Wang, C. Hein, R. Kautenburger, O. Clemens, H. Hahn, ***J. Euro Ceram Soc***, 38(2018) pp. 2318-2327, "Rare earth and transition metal based entropy stabilized perovskite type oxides."

Use of the compositions in TBC systems is not disclosed in the articles.

[0019] Metal oxides of the formula M₃O₄, where "M" represents 5 or more oxide-forming metals, which are HEOs having the spinel crystal lattice structure are disclosed in the following articles, the disclosures of each of which are incorporated herein by reference in their entireties:

19. J. Dabrowa, M. Stygar, A. Mikula, A. Knapik, K. Mroczka, W. Tejchman, M. Danielewski and M. Martin, ***Mater. Lett.*** 216(2018) pp. 32-36, "Synthesis and microstructure of (Co,Cr,Fe,Mn,Ni)₃O₄ high entropy oxide characterized by spinel structure."
20. A. Navrotsky and O. J. Klepp. a, ***J. Inorg. Nucl. Chem.***, vol 29, no. 11, pp. 2701-2714, 1967, "The thermodynamics of cation distributions in simple spinels."

Use of the compositions in TBC systems is not disclosed in the articles.

[0020] While high-entropy oxides are known, their use as a top coat in a TBC was not known. For example, co-inventor Naixie Zhou's is a co-author of the above-listed article 17 which states that "This study represents the first report of successful synthesis of high-entropy perovskite oxides (i.e. single solid-solution phases of multi-cation perovskite oxides with high configuration entropies of $\geq 1.5R$ per mole."

[0021] Gild et al, "High-entropy fluorite oxides," **Journal of the European Ceramic Society**, 38 (2018), 3578-3584, the disclosure of which is herein incorporated by reference in its entirety, is co-authored by co-inventor Tyler Harrington, and discloses eleven fluorite oxides with five principal cations (in addition to a four-principal-cation ($\text{Hf}_{0.25}\text{Zr}_{0.25}\text{Ce}_{0.25}\text{Y}_{0.25}\text{O}_{2-\delta}$) as a start point and baseline) fabricated via high-energy ball milling, spark plasma sintering, and annealing in air. Eight of the compositions, namely ($\text{Hf}_{0.25}\text{Zr}_{0.25}\text{Ce}_{0.25}\text{Y}_{0.25}\text{O}_{2-\delta}$), ($\text{Hf}_{0.25}\text{Zr}_{0.25}\text{Ce}_{0.25}\text{Y}_{0.125}\text{Yb}_{0.125}\text{O}_{2-\delta}$), ($\text{Hf}_{0.2}\text{Zr}_{0.2}\text{Ce}_{0.2}\text{Y}_{0.2}\text{Yb}_{0.2}\text{O}_{2-\delta}$), ($\text{Hf}_{0.25}\text{Zr}_{0.25}\text{Ce}_{0.25}\text{Y}_{0.125}\text{Ca}_{0.125}\text{O}_{2-\delta}$), ($\text{Hf}_{0.25}\text{Zr}_{0.25}\text{Ce}_{0.25}\text{Y}_{0.125}\text{Gd}_{0.125}\text{O}_{2-\delta}$), ($\text{Hf}_{0.2}\text{Zr}_{0.2}\text{Ce}_{0.2}\text{Y}_{0.2}\text{Gd}_{0.2}\text{O}_{2-\delta}$), ($\text{Hf}_{0.25}\text{Zr}_{0.25}\text{Ce}_{0.25}\text{Yb}_{0.125}\text{Gd}_{0.125}\text{O}_{2-\delta}$), and ($\text{Hf}_{0.2}\text{Zr}_{0.2}\text{Ce}_{0.2}\text{Yb}_{0.2}\text{Gd}_{0.2}\text{O}_{2-\delta}$), possess single-phase solid solutions of the fluorite crystal structure with high configurational entropies (on the cation sublattices), akin to those high-entropy alloys and ceramics reported in prior studies. Most high-entropy fluorite oxides (HEFOs), it is disclosed, except for the two containing both Yb and Gd, can be sintered to high relative densities. According to Gild et al, these single-phase HEFOs exhibit lower electrical conductivities and comparable hardness (even with higher contents of softer components such as Y_2O_3 and Yb_2O_3), in comparison with 8 mol. % Y_2O_3 -stabilized ZrO_2 (8YSZ). Notably, it is disclosed, these single-phase HEFOs possess lower thermal conductivities than that of 8YSZ, presumably due to high phonon scattering by multiple cations and strained lattices. The high-entropy fluorite oxides (HEFOs) consist of solid solutions with equal molar fractions of HfO_2 , ZrO_2 , and CeO_2 as the base materials, as well as the additions of the oxides of Y, Yb, Ca, Ti, La, Mg, and Gd as fluorite phase stabilizers. The thermal conductivities of YSZ, it is disclosed, have been extensively studied due to its use as a thermal barrier coating at high temperatures. The conductivity is observed to be dependent upon the porosity, fabrication method, and doping level. The measured thermal conductivities of eight of the single-phase HEFOs reported as all being lower than that of 8YSZ. However, hafnium is a very heavy metal, and the high content of hafnium in these high-entropy fluorite oxides increases the weight and density of the coating which is less desirable in aerospace applications.

[0022] The present invention provides TBC top coats with lower thermal conductivity than currently used TBC top coats, and good sintering resistance, excellent phase stability and good thermal cycling performance. The top coats are high entropy oxides which exhibit a single phase or single crystalline structure, such as tetragonal or cubic over a prolonged temperature range which may be from room temperature to the operating temperature of top coats for turbine blades in jet engines. The high entropy oxide top coats of the present invention which exhibit a single phase or maintain a phase composition without transformation throughout the heat cycling do not delaminate from a thermal bond coating or substrate at high operating temperatures due to substantial volume changes caused by a change in crystal structure or phase change. A high content of hafnium can be included, but is not required thereby lowering coating weight and density, while providing low heat conductivity and maintaining single phase crystalline structures such as cubic or tetragonal, over prolonged periods of time for unexpectedly wide temperature ranges up to and beyond top coat operating temperatures.

SUMMARY OF THE INVENTION

[0023] In accordance with the present invention, a thermal barrier coating (TBC) with low thermal conductivity which exhibits a single phase or single crystalline structure, such as tetragonal or cubic over a prolonged temperature range comprises a top coat which is a high entropy oxide (HEO) having a high configurational entropy. In aspects of the present invention, the phase composition is largely retained without transformation from room temperature to the operating temperature of top coats for gas turbine components. The HEO is of the form M_xO_y , where M represents a group of at least 5 different oxide-forming metallic cations, x represents the number of metal cations (M) or atoms, and y represents the number of oxygen anions (O) or atoms. In embodiments of the invention, at least five of the different oxide-forming metallic cations (M) may comprise: a) at least one, preferably at least two of the transition metals Sc, Y, Ti, Zr, V, Nb, Ta, Cr, Mo, W, Mn, Re, Fe, Ru, Co, Ni, Cu, or Zn, and/or b) at least one, preferably at least two, of the lanthanides La, Ce, Pr, Nd, Pm, Sm, Eu, Gd, Tb, Dy, Ho, Er, Yb, or Lu. In some embodiments at least one of the oxide-forming metallic cations comprises an alkaline-earth metal: Be, Mg, Ca, Sr, or Ba.

[0024] In embodiments of the invention the thermal barrier coating may comprise at least five different metal oxides which are:

- a) Y_2O_3 in an amount of from 5% by weight to 18% by weight,

- b) ZrO_2 in an amount of from 12% by weight to 55% by weight,
- c) CaO in an amount of from 0% by weight to 15% by weight,
- d) Gd_2O_3 in an amount of from 0% by weight to 30% by weight,
- e) La_2O_3 in an amount of from 0% by weight to 26% by weight,
- f) Yb_2O_3 in an amount of from 0% by weight to 32% by weight,
- g) TiO_2 in an amount of from 0% by weight to 10% by weight, or
- h) CeO_2 in an amount of from 0% by weight to 18% by weight,

the percentages of a) to h) adding up to at least 97% by weight and preferably between 98% and 100% by weight.

[0025] In aspects of the invention, the thermal barrier coating material used to form the top coat may be in powder form, or wire, ingot, bar or rod form. In each instance, the chemical composition of the thermal barrier coating material or top coat material may be as described for the thermal barrier coating (TBC) or top coat.

[0026] In another aspect of the invention, a thermal barrier coating (TBC) system includes a top coat and a bond coat or bond coating, with the top coat being bonded to the bond coat or bond coating. A coated substrate comprises a substrate and the thermal barrier coating system bonded to the substrate by the bond coat or coating. The thermal barrier coating system may be produced from the thermal spray powder. The thermal barrier coating system is bonded to a substrate, such as a superalloy, by the bond coat or bond coating, which is between the top coat and the substrate.

[0027] In an additional aspect of the invention, delamination of a top coat from a substrate is reduced by bonding the top coat to the substrate with the bond coat or coating. The top coat may comprise a thermal barrier coating (TBC), and the substrate may comprise a gas turbine engine component.

BRIEF DESCRIPTION OF THE DRAWINGS

[0028] The present invention is further illustrated by the accompanying drawings wherein:

Fig. 1 schematically illustrates a coated substrate having a thermal barrier coating (TBC) system which includes a top coat, such as a thermal barrier coating (TBC) bonded to the bond coat or coating in accordance with the present invention.

Fig. 2 shows the morphology (SEM micrograph) for the HEO agglomerated and sintered powder of Sample HEO 1.

Fig. 3 shows the cross section (SEM micrograph) for the HEO agglomerated and sintered powder of Sample HEO 1.

Fig. 4 shows the cross section microstructure (SEM micrograph) for the HEO thermal barrier coating, TBC top coat, of Sample HEO 1.

Fig. 5 shows the morphology (SEM micrograph) for the HEO agglomerated and sintered powder of Sample HEO 2.

Fig. 6 shows the cross section (SEM micrograph) for the HEO agglomerated and sintered powder of Sample HEO 2.

Fig. 7 shows the cross section microstructure (SEM micrograph) for the HEO thermal barrier coating, TBC top coat, of Sample HEO 2.

Fig. 8 shows the morphology (SEM micrograph) for the HEO agglomerated and sintered powder of Sample HEO 3.

Fig. 9 shows the cross section (SEM micrograph) for the HEO agglomerated and sintered powder of Sample HEO 3.

Fig. 10 shows the cross section microstructure (SEM micrograph) for the HEO agglomerated and sintered thermal barrier coating, TBC top coat, of Sample HEO 3.

Fig. 11 shows the morphology (SEM micrograph) for the HEO agglomerated and sintered powder of Sample HEO 4.

Fig. 12 shows the cross section (SEM micrograph) for the HEO agglomerated and sintered powder of Sample HEO 4.

Fig. 13 shows the cross section microstructure (SEM micrograph) for the HEO agglomerated and sintered thermal barrier coating, TBC top coat, of Sample HEO 4.

Fig. 14 shows the experimental XRD pattern for Sample HEO 4.

DETAILED DESCRIPTION OF THE INVENTION

[0029] The present invention provides a top coat, such as a thermal barrier coating (TBC) for a thermal barrier coating system which includes the top coat and at least one bond coat for bonding to a substrate, such as a superalloy employed in high temperature gas turbine engine components. There could also be multiple top coat layers on top of the at least one bond coat layer. The present invention also provides a thermal barrier coating material or top coat material for making the top coat or thermal barrier coating. The TBC top coats exhibit low thermal conductivity, good sintering resistance, excellent phase stability and good thermal cycling performance. The top coats are high entropy oxides which exhibit a single phase or single crystalline structure, such as tetragonal or cubic for prolonged periods over an unexpectedly wide temperature range which may be from room temperature to the operating temperature of top coats for turbine blades in jet engines. The high entropy oxide top coats of the present invention which exhibit a single phase throughout the heat cycling do not delaminate from a thermal bond coating or substrate at high operating temperatures due to substantial volume changes caused by a change in crystal structure or phase change and thermal internal stresses. Low coating weight and low coating density are achieved while providing low heat conductivity and maintaining single phase crystalline structures such as cubic or tetragonal, over prolonged times and unexpectedly wide temperature ranges up to and beyond top coat operating temperatures, which may be at least 1800°F, for example 2,000°F or higher, preferably at least 2300°F, or up to the melting point of the TBC top coat. In embodiments of the invention, more than one phase or crystalline structure may be present in the top coat or thermal barrier coating provided it does not adversely result in substantial volume changes caused by a change in crystal structure so as to cause delamination. While only a single phase or crystalline structure is most preferred, i.e. a phase volume of 100% (as measured for example, by X-Ray Diffraction), in embodiments of the invention where two or more phases or crystalline structures are present, the primary phase volume fraction may, for example, be at least 80%, preferably at least 90%, more preferably at least 98%.

[0030] The TBC top coats using a bond coat to bond to a substrate such as a superalloy exhibit unexpectedly high thermal cyclic fatigue resistance.

[0031] The TBC top coat, and TBC material for making the top coat is a high entropy oxide (HEO) having a high configurational entropy. In aspects of the present invention, the single phase may be retained without transformation to another phase or crystalline structure from

room temperature to the operating temperature of top coats for turbine components. The HEO is of the form M_xO_y , where M represents a group of at least 5 different oxide-forming metallic cations, x represents the number of metal cations (M) or atoms, and y represents the number of oxygen anions (O) or atoms.

[0032] In embodiments of the invention, the TBC top coat and the TBC material used to make the top coat, and the HEO is a single phase or single crystalline structure for an unexpectedly wide temperature range wherein the HEO does not undergo phase transformation to a different crystalline structure, for example, for a temperature range of at least 700°F, preferably at least 1,000°F, most preferably for at least 1,500°F prior to or up to the top coat maximum operating temperature or the melting point of the HEO. For example, in aspects of the present invention, if the maximum operating temperature in a turbine engine component is 2,000°F, then the HEO may have a single phase or single crystalline structure from 1,300°F to 2,000°F, preferably from 1,000°F to 2,000°F, most preferably from 500°F to 2,000°F, or more preferably from room temperature to 2000°F or higher, preferably to at least 2300°F, such as to an HEO melting point of 2400°F. In aspects of the invention, the HEO may be a single phase or single crystalline structure over or throughout an 800°F temperature range extending from 1800°F to 2,600°F, or extending from 1,300°F to 2,100°F. The wider the temperature range without a phase change, the better, because, for example, the number of phases changes would be fewer as the top coat cycles up to and back down from the turbine operating temperature, or as the operating temperature fluctuates, thereby helping to reduce thermal expansion and contraction and thermal stresses.

[0033] In preferred aspects of the invention, the TBC top coat, TBC coating material, and the HEO may have only a single phase or single crystalline structure which is only, for example, cubic or tetragonal from room temperature up to at least 1800°F, preferably up to at least 2,000°F, more preferably, up to at least 2300°F, for example from room temperature up to the melting point of the HEO. In embodiments of the invention, the HEOs have a melting point of at least 1,150°C (2,102°F), preferably at least 1,300°C (2,372°F), more preferably at least 1,315°C (2,399°F).

[0034] In embodiments of the invention the inherent thermal conductivity at 25°C of the HEO topcoat or coating may be less than 1.5 (W/m-K) preferably less than 1.2 (W/m-K), and more preferably less than 0.9W/m-K. In embodiments of the invention, the HEO coating may have a density lower than the theoretical density (i.e. may contain porosity) thus reducing the

thermal conductivity of the HEO topcoat coating to less than 1.3 (W/m-K) preferably less than 1.0 (W/m-K) and more preferably less than 0.8 (W/m-K). This is demonstrated in Table 2.

[0035] In embodiments of the invention the TBC top coat or coating Archimedes density may be less than 7g/cm^3 , for example from 5g/cm^3 , to 6.5g/cm^3 , preferably less than 6.3g/cm^3 , for example from 5.25g/cm^3 to 6.25g/cm^3 , more preferably less than 6.0, for example 5.30g/cm^3 to 5.90g/cm^3 .

[0036] In embodiments of the invention, the oxide ceramic, or HEO intended to be used as a thermally insulating material or thermal barrier coating, may have an overall combined atomic composition of which can be expressed as M_xO_y where M represents a group of at least 5 different oxide-forming metallic cations and where the configurational entropy S_{CONFIG} of the oxide is $1.5R$ per mole or greater, where R is the gas constant $8.314\text{ J.K}^{-1}.\text{mol}^{-1}$; this definition using the value of S_{CONFIG} being a commonly-accepted definition of a high-entropy material. The metal cations "M" and oxygen anions "O" may be distributed on one or more crystal sub-lattices. In aspects of the present invention, the TBC top coat may have a configurational entropy S_{config} of the oxide which is below $1.5R$ per mole, for example $1.0 R$ per mole or greater, or $1.3R$ per mole or greater provided the heat conductivity is low and the metal oxide maintains phase composition for an unexpectedly wide temperature range, as discussed above, wherein the metal oxide does not undergo phase transformation and the primary phase volume fraction maintains, for example, at least 80%, preferably at least 90%, more preferably at least 98% and the melting point is above the operating temperature of the coating as described above.

[0037] M_xO_y is standard metallurgical shorthand. For example, the carbide $(\text{Cr}, \text{Mo}, \text{W}, \text{Fe})_{23}\text{C}_6$ is commonly referred to as $M_{23}\text{C}_6$. and in the same way M_xO_y may be used to describe the oxide $(\text{Zr}, \text{Ce}, \text{Y}, \text{Yb}, \text{Gd}, \text{Dy})_x\text{O}_y$ where "M" represents 5 or more oxide-forming metals.

[0038] In embodiments of the present inventions, these metals "M" may preferably be selected from the group of non-toxic and non-radioactive oxide-forming metals, such as:

Transition Metals:

- Sc, Y

- Ti, Zr, Hf
- V, Nb, Ta
- Cr, Mo, W
- Mn, Re
- Fe, Ru, Co, Ni, Cu, Zn, and

Lanthanides:

La, Ce, Pr, Nd, Pm, Sm, Eu, Gd, Tb, Dy, Ho, Er, Yb, Lu.

In some embodiments of the invention at least one alkaline-earth metal may preferably be selected, such as

- Be, Mg, Ca, Sr, Ba

[0039] In embodiments of the invention, the following metals are more preferred for use in HEO TBCs:

Transition Metals:

- Y
- Ti, Zr, Hf, V
- Cr, Mo, W

Lanthanides:

- La, Ce, Pm, Sm
- Eu, Gd
- Tb
- Dy, Er, Yb

In some embodiments of the invention at least one alkaline-earth metal is more preferably selected, such as

- Mg, Ca

[0040] In embodiments of the invention, at least one, preferably at least two of the transition metals, and/or at least one, preferably at least two of the lanthanides may be employed in the at least five of the different oxide-forming metallic cations (M).

[0041] While hafnium (Hf) has a very high melting point, in embodiments of the invention, it may be eliminated or used in low amounts, for example less than 2.0% by weight, preferably less than 1% by weight. Although higher amounts of hafnium may be employed, for example up to 15% by weight or more, high contents of hafnium are not necessarily required, thereby allowing lowering of the coating weight and density, while providing low heat conductivity and maintaining single phase crystalline structures such as cubic or tetragonal, over prolonged periods of time for unexpectedly wide temperature ranges up to and beyond top coat operating temperatures.

[0042] Still further, the metal cations "M" and oxygen anions "O" may be distributed on one or more crystal sub-lattices. This means that the oxide, such as the exemplary oxide $(\text{Zr,Ce,Y,Yb,Gd,Dy})_x\text{O}_y$, may be physically manifested as one combined oxide structure of as-yet-unknown crystallography $(\text{Zr,Ce,Y,Yb,Gd,Dy})_x\text{O}_y$, or it may partition itself into 2 (or more) more-commonly-known crystal lattices e.g. $(\text{Y,Yb,Gd,Dy})_2\text{O}_3$ and $(\text{Zr,Ce})\text{O}_2$. Accordingly, in the latter case this would imply 2 atoms from the group (Y,Yb,Gd,Dy) for every 3 oxygen atoms, and 1 atom from the group Zr,Ce for every 2 oxygen atoms, within the overall composition. With intimate mixing of these oxide lattices it may not necessarily be able to detect separate phases in the HEO structure when examined by scanning electron microscopy.

[0043] In aspects of the present invention, known high entropy oxides, such as those of the above-discussed and listed references incorporated by reference herein in their entireties, may be employed as a top coat provided they are primarily single phase, or single crystal structure such as tetragonal or cubic, provided the heat conductivity is low and the metal oxide maintains phase composition for an unexpectedly wide temperature range, as discussed above, and the melting point is above the operating temperature of the coating as described above. Most preferably, the HEO should not undergo significant transformation or alter phase fraction from room temperature to the operating temperature or melting point of the top coat, as described above.

[0044] In accordance with the present invention, in embodiments the thermal barrier coating may include a top coat, wherein the top coat is a high entropy oxide (HEO) having a high

configurational entropy, the HEO being of the form M_xO_y , where M represents a group of at least 5 different oxide-forming metallic cations, x represents the number of metal cations (M) or atoms, and y represents the number of oxygen anions (O) or atoms, the HEO is a single phase across the operating temperature range of the top coat, and at least five of the different oxide-forming metallic cations (M) may comprise:

- a) at least one of the alkaline-earth metals or Group II of the Periodic Table, Be, Mg, Ca, Sr, or Ba, and/or
- b) at least one of the transition metals Sc, Y, Ti, Zr, V, Nb, Ta, Cr, Mo, W, Mn, Re, Fe, Ru, Co, Ni, Cu, or Zn, and/or
- c) at least one of the lanthanides La, Ce, Pr, Nd, Pm, Sm, Eu, Gd, Tb, Dy, Ho, Er, Yb, or Lu.

[0045] In preferred embodiments of the thermal barrier coating the at least five of the different oxide-forming metallic cations (M) may comprise:

- a) at least one of the transition metals Y, Ti, Zr, V, Cr, Mo, or W, and/or
- b) at least one of the lanthanides La, Ce, Pm, Sm, Eu, Gd, Tb, Dy, Er, or Yb, and
- c) in some embodiments at least one alkaline-earth metal: Mg, or Ca.

More preferably, the at least five of the different oxide-forming metallic cations (M) of the thermal barrier coating comprise at least five of Y, Zr, Ca, Gd, La, Yb, Ti, or Ce.

[0046] In aspects of the present invention the thermal barrier coating and the thermal barrier coating material or top coat material may comprise at least five different metal oxides which are:

- a) Y_2O_3 in an amount of from 5% by weight to 20% by weight, preferably from 8% by weight to 18% by weight,
- b) ZrO_2 in an amount of from 12% by weight to 55% by weight, preferably from 17% by weight to 52% by weight,
- c) CaO in an amount of from 0% by weight to 15% by weight, preferably from 0% by weight to 11% by weight,
- d) Gd_2O_3 in an amount of from 0% by weight to 30% by weight, preferably from 0% by weight to 28% by weight,
- e) La_2O_3 in an amount of from 0% by weight to 26% by weight, preferably from 0% by weight to 24% by weight,

- f) Yb_2O_3 in an amount of from 0% by weight to 32% by weight, preferably from 0% by weight to 30% by weight,
- g) TiO_2 in an amount of from 0% by weight to 10% by weight, preferably from 0% by weight to 7% by weight, or
- h) CeO_2 in an amount of from 0% by weight to 18% by weight, preferably from 0% by weight to 15% by weight,

the percentages of a) to h) adding up to 100% by weight. In embodiments, the thermal barrier coating and the thermal barrier coating material or top coat material may further include at least one additional metal oxide, such as HfO_2 , SiO_2 , MgO , or Al_2O_3 , as melting point adjusters, stabilizers, dopants, or impurities, in an amount up to 3% by weight, for example, up to 2% by weight or less than or equal to 1% by weight, the weight percentages of all of the oxides adding up to 100% by weight. In embodiments of the invention, the percentages of the five metal oxides may add up to at least 97% by weight, for example at least 99% by weight, and the additional metal oxide or oxides may be present in an amount of up to 3% by weight, or up to 1% by weight.

[0047] In more preferred embodiments, the thermal barrier coating and the thermal barrier coating material or top coat material may comprise at least five different metal oxides which include: Y_2O_3 in an amount of from 8% by weight to 12% by weight, ZrO_2 in an amount of from 48% by weight to 55% by weight, Yb_2O_3 in an amount of from 14% by weight to 18% by weight, TiO_2 in an amount of from 4% by weight to 8% by weight, and CeO_2 in an amount of from 12% by weight to 17% by weight, said percentages adding up to 100% by weight. Hafnium, and additional other metal oxides may optionally be included in an amount up to 2% by weight, for example, less than or equal to 1% by weight, the weight percentages of all of the oxides adding up to 100% by weight.

[0048] In other preferred embodiments, the thermal barrier coating and the thermal barrier coating material or top coat material may comprise at least five different metal oxides which include:

three metal oxides which are

- Y_2O_3 in an amount of from 13% by weight to 19% by weight,
 - ZrO_2 in an amount of from 14% by weight to 25% by weight, and
 - Gd_2O_3 in an amount of from 20% by weight to 30% by weight, and
- any two of the metal oxides which are

Yb₂O₃ in an amount of from 23% by weight to 32% by weight,
La₂O₃ in an amount of from 18% by weight to 25% by weight, or
CaO in an amount of from 6% by weight to 12% by weight,
the percentages of the five metal oxides adding up to at least 97% by weight and preferably between 98% and 100% by weight.

[0049] In embodiments of the invention, the TBC top coat material or HEO may be manufactured in powder form or bulk form, such as wire, bar, rod or ingot form. The TBC top coat material powder may be a homogeneous mixture of separate powders of each of the components of the TBC top coat material. The TBC top coat material powder may also be composed of particles, each of which contain all or some of the components of the bond coating material. For example, a bulk form of all the components of the TBC top coat material or HEO may be ground to obtain a powder. The particle size of the TBC top coat material may depend upon the coating method employed. Conventional particle size distributions conventionally employed with a given coating method may be used with the TBC top coat materials or HEOs of the present invention.

[0050] The bond coating material may be any conventional or known bond coat material such as those used for the coating of gas turbine engine parts, such as known MCrAlY bonding layers for bonding a TBC top coat to a substrate, such as a superalloy. By way of example, M may represent Ni, Co, Fe or their combinations. To increase the performance of a bond coat, Hf, Re and Pt and various other rare earth elements may often be added in advanced bond coats. Non-limiting examples of the bond coat material which may be employed include those disclosed in U.S. Patent No. 4,117,179 to Jackson et al, U.S. Patent No. 5,141,821 to Lugscheider, and U.S. Patent No. 4,275,124 to McComas et al, the disclosures of which are each herein incorporated by reference in their entireties.

[0051] The substrate may be any known or conventional material or article in need of a top coat or barrier coating (TBC). Non-limiting examples of the substrate include alloys, or superalloys used in the manufacture of gas turbine engine parts, such as Hastelloy® X, as disclosed in U.S. Patent No. 4,124,737 to Wolfa et al, the disclosure of which is herein incorporated by reference in its entirety. Hastelloy®X, as disclosed by Wolfa et al, has a nominal composition of 22.0 wt. % chromium; 9.0 wt.% molybdenum, 18.5 wt.% iron; 1.5 wt.% cobalt; 0.6wt. % tungsten, 1.0wt% silicon, 1.0 wt. % manganese, 0.1 wt. % carbon, and the balance nickel. Other non-limiting examples of known and conventional substrates which

may be coated with the TBC top coat of the present invention include steel, stainless steel, other iron-base alloys with low alloying content, chromium and chromium base alloys, and the refractory metal and refractory metal base alloys. Non-limiting examples of superalloy substrates which may be coated with the TBC top coat of the present invention are known carbide reinforced superalloys, such as nickel-base and cobalt-base superalloys, directionally solidified nickel-base and cobalt-base superalloys including eutectic alloys, as well as refractory alloys, etc. as disclosed in U.S. Patent No. 4,117,179, the disclosure of which is hereby incorporated by reference in its entirety. Non-limiting examples of substrates or articles which may be coated with the TBC top coat of the present invention include components used in the hot sections of gas turbines and various jet engine components.

[0052] In another aspect of the invention, as schematically illustrated in FIG. 1, a thermal barrier coating (TBC) system 1 includes a top coat 2 and the bond coat or bond coating 3, with the top coat 2, such as a TBC, being bonded to the bond coat or bond coating 3 at the top coat/ bond coat interface 5. A coated substrate 10 comprises a substrate 15 and the thermal barrier coating system 1 bonded to the substrate 15 by the bond coat or coating 3 at the substrate/bond coat interface 20. The thermal barrier coating system 1 may be produced from the thermal spray powder. The thermal barrier coating system 1 is bonded to a substrate 15, such as a superalloy or gas engine turbine component, by the bond coat or bond coating 3, which is between the top coat 2 and the substrate 15.

[0053] In embodiments of the invention, multiple bond coats or bond coatings 3 and multiple top coats 2 may be employed, with each top coat 2 being on top of a bond coat 3 in alternating fashion, to provide a plurality of TBC systems 1, stacked and bonded one on another, with the bottommost bond coat 3 bonded to the substrate 15.

[0054] In an additional aspect of the invention, a method is provided for reducing delamination of a top coat from a substrate by bonding the TBC top coat having a single phase or single crystalline structure, such as a tetragonal or cubic crystalline structure to the substrate with a bond coat or coating.

[0055] The TBC top coat, or HEO, and the bond coat or bond coating or bonding layer, may be deposited, applied, or laminated on the substrate using conventional thermal spray processes, such as air plasma spray, suspension plasma, high velocity oxy-fuel spray (HVOF), low pressure plasma spray (LPPS), vacuum plasma spray (VPS), chemical vapor deposition (CVD), plasma physical vapor deposition (PS-PVD), physical vapor deposition (PVD) which

includes vacuum deposition methods, such as sputtering and evaporation, and conventional flame spray processes, such as combustion wire spray, and combustion powder spray, electric arc wire spray, powder flame spray, and electron beam physical vapor deposition (EBPVD) . Conventional and known coating layer thicknesses may be employed for the bond coat or coating, and the top coat or TBC or HEO.

[0056] In embodiments of the invention, a thermal barrier coating material or the HEO material may be manufactured by agglomerating and sintering a thermal barrier coating material to obtain a sintered agglomerate, and forming the sintered agglomerate into a powder for thermal spraying using known techniques and processes. Agglomerating and sintering is a specifically novel way to manufacture HEOs. The process is advantageous because it: 1) reduces the diffusion pathways, and 2) is more amenable to industrial production. The reduction of the diffusion pathways is very advantageous as homogenizing high temperature materials into a single phase can be expensive and time consuming when compared with other processes. In embodiments of the invention, a thermal barrier coating may be manufactured by agglomerating and sintering a thermal barrier coating material to obtain a sintered agglomerate, and forming the sintered agglomerate into a powder for thermal spraying using known process parameters and techniques.

[0057] The present invention is further illustrated by the following non-limiting examples where all parts, percentages, proportions, and ratios are by weight, all temperatures are in °C, and all pressures are atmospheric unless otherwise indicated.

EXAMPLES

[0058] The compositions of four HEO TBC top coat materials (as powders), of the present invention used to make HEO TBC top coats of the present invention for a coated substrate are shown in Table 1.

[0059] The Table 1 compositions of the samples tested, and the results of the testing are:

TABLE 1: Compositions of the HEO TBC Top Coat Materials
Used To Make HEO TBC Top Coats For A Coated Substrate

Sample	ZrO ₂ wt %	Y ₂ O ₃ wt %	Gd ₂ O ₃ wt %	Yb ₂ O ₃ wt %	La ₂ O ₃ wt %	CaO wt %	CeO ₂ wt %	TiO ₂ wt %	Traces wt %
HEO 1	23.4	16.0	26.0		23.5	10.5			0.6
HEO 2	18.3	17.0	27.1	29.4		7.9			0.3
HEO 3	16.4	14.9	23.3	25.3	20.0				0.1
HEO 4	51.9	9.9		17.1			14.6	6.0	0.5

[0060] For the TBC top coat materials (HEOs): the coating density, relative density, coating phase, and thermal conductivity measured at 25°C, are shown in Table 2.

[0061] The Table 2 results of the properties for the samples tested are:

TABLE 2: Properties of the HEO TBC Top Coats

Property	Units Measurement	HEO 1	HEO 2	HEO 3	HEO 4
Archimedes Density	g/cm ³	5.36	5.72	6.23	5.87
Relative Density	%	91.8	92.3	92.7	93.1
Primary Phase Fraction from XRD (Vol%)	Vol %	65%	87%	>98%	>99%
Thermal conductivity at 25°C	W/m-K	1.22	1.30	1.21	0.77

[0062] The powder morphology (SEM micrograph), powder cross section (SEM micrograph), and coating microstructure (SEM micrograph) for the HEO agglomerated and sintered thermal barrier coatings of samples HEO 1, HEO 2, HEO 3 and HEO 4 are shown in Figs. 2-13. An example of the X-Ray diffraction pattern of one of the HEO compounds, HEO 4, is shown in Fig 14.

[0063] Further, at least because the invention is disclosed herein in a manner that enables one to make and use it, by virtue of the disclosure of particular exemplary embodiments, such as for simplicity or efficiency, for example, the invention can be practiced in the absence of any step, additional element or additional structure that is not specifically disclosed herein.

[0064] It is noted that the foregoing examples have been provided merely for the purpose of explanation and are in no way to be construed as limiting of the present invention. While the present invention has been described with reference to an exemplary embodiment, it is understood that the words which have been used herein are words of description and illustration, rather than words of limitation. Changes may be made, within the purview of the appended claims, as presently stated and as amended, without departing from the scope and spirit of the present invention in its aspects. Although the present invention has been described herein with reference to particular means, materials and embodiments, the present invention is not intended to be limited to the particulars disclosed herein; rather, the present invention extends to all functionally equivalent structures, methods and uses, such as are within the scope of the appended claims.

What is claimed is:

1. A thermal barrier coating comprising:

a top coat, wherein the top coat is a high entropy oxide (HEO) having a high configurational entropy, the HEO being of the form M_xO_y , where M represents a group of at least 5 different oxide-forming metallic cations, x represents the number of metal cations (M) or atoms, and y represents the number of oxygen anions (O) or atoms,

wherein the HEO maintains phase composition without transformation from room temperature to the operating temperature range of the top coat, and at least five of the different oxide-forming metallic cations (M) comprise Y, Zr, Ca, Gd, La, Yb, Ti, or Ce, such that the HEO comprises at least five different metal oxides which are from the group:

- a) Y_2O_3 in an amount of from 5% by weight to 20% by weight,
- b) ZrO_2 in an amount of from 12% by weight to 55% by weight,
- c) CaO in an amount of from 0% by weight to 15% by weight,
- d) Gd_2O_3 in an amount of from 0% by weight to 30% by weight,
- e) La_2O_3 in an amount of from 0% by weight to 26% by weight,
- f) Yb_2O_3 in an amount of from 0% by weight to 32% by weight,
- g) TiO_2 in an amount of from 0% by weight to 10% by weight, or
- h) CeO_2 in an amount of from 0% by weight to 18% by weight,

wherein the percentages of the at least five of the metal oxides add up to at least 97% by weight.

2. The thermal barrier coating of claim 1 comprising at least five different metal oxides from the group:

- a) Y_2O_3 in an amount of from 8% by weight to 18% by weight,
- b) ZrO_2 in an amount of from 17% by weight to 52% by weight,
- c) CaO in an amount of from 0% by weight to 11% by weight,
- d) Gd_2O_3 in an amount of from 0% by weight to 28% by weight,
- e) La_2O_3 in an amount of from 0% by weight to 24% by weight,

f) Yb_2O_3 in an amount of from 0% by weight to 30% by weight,
g) TiO_2 in an amount of from 0% by weight to 7% by weight, or
h) CeO_2 in an amount of from 0% by weight to 15% by weight,
wherein the percentages of the at least five metal oxides add up to at least 97% by weight.

3. The thermal barrier coating of claim 1 comprising Y_2O_3 in an amount of from 8% by weight to 12% by weight, ZrO_2 in an amount of from 48% by weight to 55% by weight, Yb_2O_3 in an amount of from 14% by weight to 18% by weight, TiO_2 in an amount of from 4% by weight to 8% by weight, and CeO_2 in an amount of from 12% by weight to 17% by weight, said percentages adding up to at least 97% by weight.

4. The thermal barrier coating of claim 3 further comprising at least one additional metal oxide in an amount up to 2% by weight, said weight percentages adding up to at least 99% by weight.

5. The thermal barrier coating of claim 3 further comprising HfO_2 in an amount up to 2% by weight, said weight percentages adding up to at least 99% by weight.

6. The thermal barrier coating of claim 1 comprising:

three metal oxides which are

Y_2O_3 in an amount of from 13% by weight to 19% by weight,

ZrO_2 in an amount of from 14% by weight to 25% by weight, and

Gd_2O_3 in an amount of from 20% by weight to 30% by weight, and

any two of the metal oxides which are

Yb_2O_3 in an amount of from 23% by weight to 32% by weight,

La_2O_3 in an amount of from 18% by weight to 25% by weight, or

CaO in an amount of from 6% by weight to 12% by weight,

the percentages of the five metal oxides adding up to at least 97% by weight.

7. The thermal barrier coating of claim 1, wherein the configurational entropy of the oxide is at least $1.5R$ per mole, where R is the gas constant $\text{J}\cdot\text{K}^{-1}\cdot\text{mol}^{-1}$.

8. The thermal barrier coating of claim 1, wherein the five or more different oxide-forming metallic cations are present in compositions between 5 and 35 atomic %.

9. The thermal barrier coating of claim 1, further comprising a thermal barrier coating bond coat.

10. The thermal barrier coating of claim 1, wherein the HEO is a single phase or single crystalline structure from room temperature up to at least 2,000°F.

11. The thermal barrier coating of claim 1, wherein the HEO is a single phase or single crystalline structure which is cubic from room temperature up to at least 2,000°F.

12. The thermal barrier coating of claim 1, wherein the HEO is a single phase or single crystalline structure from 1800°F to 2600°F.

13. The thermal barrier coating of claim 1, wherein the HEO is a single phase or single crystalline structure which is tetragonal from room temperature up to at least 2000°F.

14. The thermal barrier coating of claim 1, wherein the HEO is a single phase or single crystalline structure from room temperature up to the melting point of the HEO.

15. The thermal barrier coating of claim 1, wherein the HEO does not undergo phase transformation to a different crystalline structure for at least 700°F up to the melting point of the HEO.

16. A thermal barrier coating material comprising:

a high entropy oxide (HEO) having a high configurational entropy, the HEO being of the form M_xO_y , where M represents a group of at least 5 different oxide-forming metallic cations, x represents the number of metal

cations (M) or atoms, and y represents the number of oxygen anions (O) or atoms,

wherein the HEO is a single phase across the operating temperature range of a top coat, and at least five of the different oxide-forming metallic cations (M) comprise Y, Zr, Ca, Gd, La, Yb, Ti, or Ce, such that the HEO comprises at least five different metal oxides which are from the group:

- a) Y_2O_3 in an amount of from 5% by weight to 20% by weight,
- b) ZrO_2 in an amount of from 12% by weight to 55% by weight,
- c) CaO in an amount of from 0% by weight to 15% by weight,
- d) Gd_2O_3 in an amount of from 0% by weight to 30% by weight,
- e) La_2O_3 in an amount of from 0% by weight to 26% by weight,
- f) Yb_2O_3 in an amount of from 0% by weight to 32% by weight,
- g) TiO_2 in an amount of from 0% by weight to 10% by weight, or
- h) CeO_2 in an amount of from 0% by weight to 18% by weight,

wherein the percentages of the at least five of the metal oxides add up to at least 97% by weight.

17. The thermal barrier coating material of claim 16 comprising Y_2O_3 in an amount of from 8% by weight to 12% by weight, ZrO_2 in an amount of from 48% by weight to 55% by weight, Yb_2O_3 in an amount of from 14% by weight to 18% by weight, TiO_2 in an amount of from 4% by weight to 8% by weight, and CeO_2 in an amount of from 12% by weight to 17% by weight, said percentages adding up to at least 97% by weight.

18. The thermal barrier coating material of claim 16 which is at least one of a powder, a wire, a bar, an ingot and a rod.

19. A thermal spray powder comprising the top coat material of claim 16.

20. A coated substrate comprising a substrate and a thermal barrier coating of claim 9 wherein the top coat of the thermal barrier coating is bonded by the thermal barrier coating bond coat to the substrate.

21. A method for reducing delamination of a top coat from a substrate comprising bonding a thermal barrier coating of claim 9 to a substrate wherein the top coat of the thermal barrier coating is bonded by the thermal barrier coating bond coat to the substrate.

22. The thermal barrier coating material of claim 16, which is agglomerated and sintered.

23. The thermal barrier coating material of claim 22 which is in powdered form.

24. A method for making a thermal barrier coating comprising agglomerating and sintering a thermal barrier coating material of claim 16 to obtain a sintered agglomerate, and forming the sintered agglomerate into a powder for thermal spraying.