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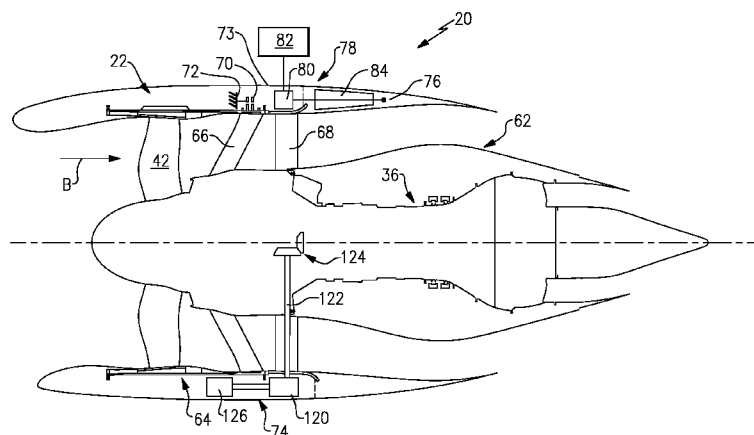


FIG.2

(57) Abstract: A gas turbine engine thrust reverser system includes a fan nacelle having a fixed portion and a movable portion configured to move relative to the fixed portion between opened and closed positions in response to a command to selectively provide an opening between the fixed and movable portions. A fan case is arranged within the fan nacelle. The fan case includes an integral bull nose adjacent to the opening in the open position and to the movable portion in the closed position. The bull nose has a curved wall sloping radially outward.

