

IN SITU ENGINE COMPONENT REPAIR

ABSTRACT

A method for repairing a turbine engine (10) includes determining a measure of a gap (86) between the outer ends (76) of a plurality of airfoils (60) and an outer band (80) is greater than a predetermined amount; and depositing an abradable material on the outer band (80) in situ to reduce the measure of the gap (86) between the outer ends (76) of the plurality of airfoils (60) and the outer band (80).

[Figure 3]