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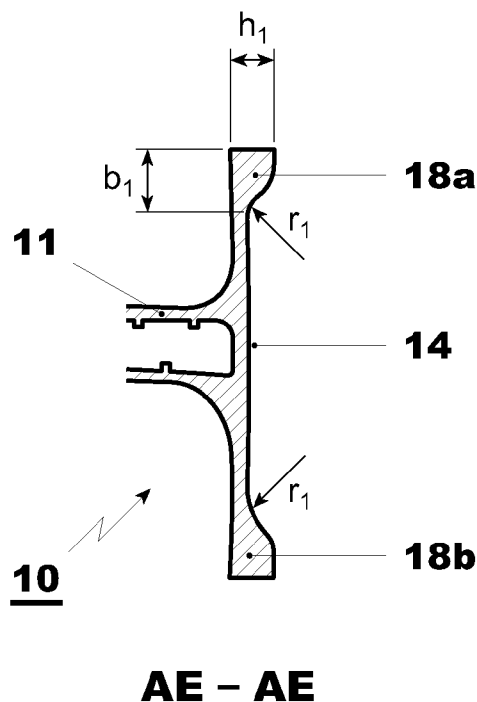
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(54) Title: BLADE FOR A GAS TURBINE

FIG. 5



(57) Abstract: A blade (10) for a gas turbine comprises an airfoil (11), which extends along a longitudinal axis (21) from a blade root (20) to a blade tip (12), and has a shroud segment (14) at said blade tip (12), which shroud segment (14) abuts with first and second edges against similar shroud segments of adjacent blades to make up a ring-like shroud, whereby said first and second edges are each provided with a respective side rail (18, 19) on the upper side of said shroud segment (14). To optimize the mechanical and thermal properties of the shroud segment (14), each of said side rails is subdivided into sections (18, 19) of different height and/or width.



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