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(71) Applicant: UNITED TECHNOLOGIES CORPORATION [US/US]; One Financial Plaza, Hartford, Connecticut 06101 (US).

(72) Inventors: XU, JinQuan; 11 LedgeWood Road, Groton, Connecticut 06340 (US). LEVASSEUR, Glenn; 51 Nelkin Road, Colchester, Connecticut 06415 (US).

(74) Agents: DE RUYTER, Matthew, J. et al.; 312 South Third Street, Minneapolis, Minnesota 55415-1028 (US).

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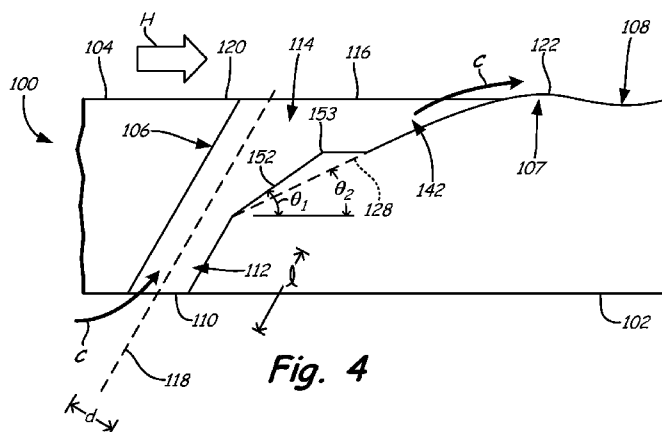


Fig. 4

(57) Abstract: A gas turbine engine component includes a wall having first and second wall surfaces, a cooling hole extending through the wall and a convexity. The cooling hole includes an inlet located at the first wall surface, an outlet located at the second wall surface, a metering section extending downstream from the inlet and a diffusing section extending from the metering section to the outlet. The diffusing section includes a first lobe diverging longitudinally and laterally from the metering section and a second lobe adjacent the first lobe and diverging longitudinally and laterally from the metering section. The convexity is located near the outlet.