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METHOD FOR REDUCING VOID VOLUMES IN CAVITIES FOR LASER DRILLING

5 TECHNICAL FIELD

This invention generally relates to a method for laser drilling a structure having a cavity, the method including steps related to filling the cavity with a blocking material to protect the interior of the cavity from the laser beam. More particularly, this invention relates to a method for disposing blocking material in
10 the cavity and minimizing the formation of voids in the blocking material.

BACKGROUND OF THE INVENTION

Airfoils for gas turbine engines are disposed in a flowpath for working medium gases. Examples of such airfoils are turbine blades and turbine vanes.
15 The airfoils are bathed in hot gases as the gases are flowed through the engine. Cooling air is flowed through passages on the interior of the airfoil under operative conditions to keep the temperature of the airfoil, such as a turbine vane or turbine blade, within acceptable limits.

In addition, the airfoil may have cooling air holes extending from the
20 interior to the exterior of the airfoil. The cooling air holes duct cooling air from passages on the interior of the airfoil through the hot walls to the exterior. The exhausted cooling air provides transpiration cooling as the air passes through the wall and film cooling with a film of cooling air on the exterior as the air is discharged from the airfoil. The film of cooling air provides a barrier between the
25 airfoil and the hot, working medium gasses.

The cooling air holes are small and may have diameters that are in a range of eleven to seventeen mils (.011-.017 inches). The holes are drilled in pre-determined patterns and are contoured to insure adequate cooling of the airfoil.

30 One way to drill the holes uses a laser to direct a beam of coherent energy at the exterior of the airfoil. The intense radiation from the laser beam burns



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through the wall of the airfoil, leaving behind a hole which provides a satisfactory conduit for cooling air. As the laser beam penetrates through the airfoil wall into an interior cavity, the laser beam may strike adjacent structure on the other side of the cavity causing unacceptable damage to the airfoil. Accordingly, blocking material may be disposed in the cavity to block the laser beam from striking walls bounding the cavity after the beam penetrates through the airfoil wall.

One approach is to leave disposed within the airfoil the ceramic casting core around which the blade is poured during the manufacturing process. The ceramic core provides a suitable blocking material. The ceramic core is subsequently removed by well known leaching techniques. This approach is described in U.S. Patent 5, 222, 617 entitled "Drilling Turbine Blades" issued to Gregore, Griffith and Stroud. However, the presence of the core after casting prevents initial inspection of the interior of the airfoil. The ceramic material may also be difficult to remove once the cooling air holes are drilled. In addition, the core is not available during repair processes for the airfoil which may require redrilling of the cooling air holes.

Another example of a blocking material is wax or a wax-like material. The material is melted so that it may easily flow into interior passages, such as the leading edge passage of the airfoil. The temperature of the molten material above its melting point, may exceed two hundred and fifty degrees Fahrenheit (250°). The molten material may be poured or injected into the cavity or may even be sprayed or painted on the surface to be protected. However, the molten material may severely scald personnel working with the material. In addition, the wax may extend between two closely adjacent cooling air holes. The wax adjacent the first hole, which blocks the laser beam as the second hole is drilled, may melt as the first hole is drilled by the laser beam. This causes a void to form in the wax. As a result, the energy from the laser beam at the second hole may not be sufficiently dissipated by the wax as it passes through the portion of the passage having the void. Damage may occur to the airfoil as the second

hole is drilled because the beam, after it penetrates through the wall at the second hole, may strike the interior wall of the airfoil.

One wax-like blocking material which uses an additive to avoid forming voids is discussed in U.S. Patent 5,049,722, issued to Corfe and Stroud, entitled
5 "Laser Barrier Material And Method Of Laser Drilling." In Corfe, a PTFE (polytetra fluoroethylene) wax-like material is disposed in a wax base. The PTFE helps avoid the formation of voids. Disposing such material on the interior of a leading edge passage is particularly difficult for some airfoils. Often the leading edge passage has no connection during fabrication with the exterior of
10 the airfoil. It is a blind or dead end passage prior to the drilling operation except for small impingement holes which place the passage in gas communication with an adjacent passage. The adjacent passage also has an opening for receiving cooling air which is flowed to the leading edge passage. Accordingly, personnel must carefully pour the molten material in the inlet opening and manipulate the
15 airfoil to avoid bubbles in the material in the leading edge passage.

Still another approach is to use a masking agent, such as an epoxy resin, which is disposed in the airfoil in a fluid state. The epoxy resin is disposed in the airfoil by simply pouring the resin into the airfoil. The epoxy resin is at room temperature and poses no scalding hazard to personnel. The epoxy resin is
20 further processed to harden the fluid and cause it to become a more solid material similar to the PTFE wax mentioned in U.S. Patent 5, 049,722. However, the resin is relatively viscous compared to molten wax and has difficulty in flowing through small connecting passages on the interior of the airfoil.

As a result, air bubbles tend to form in blocking material in blind cavities,
25 such as the leading edge passage, and in passages which are blocked by the in flowing viscous material. These air bubbles result from not flowing enough epoxy resin into the leading edge region to eliminate the voids. In addition, the structural design of the airfoil may cause the airfoil to trap an air bubble with structure within the leading edge passage.

30 Accordingly, scientists and engineers working under the direction of Applicants' Assignee have sought to develop a method for disposing fluid

blocking material in a structure having a shaped cavity such as the leading edge of an airfoil such that the fluid material in the leading edge cavity is hardened without forming voids in the cavity that would cause unacceptable damage to structure bounding the cavity during laser drilling.

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SUMMARY OF INVENTION

According to the present invention, a method for disposing a hardenable blocking material in a shaped cavity having trapped pockets of gas prior to laser drilling through the region of the pocket includes 1) expanding the volume of the trapped gas in the hardenable material to force part of the trapped gas out of the pocket and to increase buoyant forces acting on the forced out gas and, 2) compressing the remaining trapped gas in the pocket to reduce the volume of the pocket as the material is hardened.

In accordance with the present invention, the steps of expanding and compressing the trapped gas include varying the temperature or pressure or both of the external atmosphere to vary the volume of the trapped gas.

In accordance with one detailed embodiment of the present invention, the method for disposing hardenable material in a shaped cavity includes forming a vent passage adjacent the region in which a pocket of gases might form, expanding any gases in the pocket until at least a portion of the trapped gases are in flow communication with the vent passage, and allowing a portion of the expanded gases to escape through the vent passage.

A primary feature of the present method is the expansion of the volume of the trapped gas by adjusting the pressure of the environment acting on the gas. Another feature is the step of drilling the cooling air hole which takes place through a region from which the gas has been forced out by compressing the gas. Still another feature is disposing a vent passage in close proximity to the location through which the laser beam penetrates such that expansion of the gas at the partial vacuum pressure causes a sufficient quantity of gas to escape such that the pressure on the gas as the blocking material is hardened moves the volume of gas away from the line of penetration of the laser beam.

A primary advantage of the present invention is the level of cost of drilling holes in an article with a laser which results from being able to drill such holes in regions of the article that had voids from trapped gas in the blocking material by decreasing the size of the voids. Another advantage of the present invention is the cost and ease of disposing the blocking material in blind cavities on the interior of an article which results from avoiding the need for manipulating the article to remove trapped gas bubbles from the article. Still another advantage of the present invention is the health and safety of personnel which results from use of higher viscosity substances such as epoxy resin at reasonable temperatures during the filling operation which avoids scalding of personnel working with such material.

BRIEF DESCRIPTION OF DRAWINGS

Fig. 1 is a side elevation view of a turbine airfoil of the turbine vane type with part of the flow directing portion of the airfoil broken away and sectioned to show the interior of the airfoil.

Fig. 2 is a cross sectional view of the partially formed stator vane shown in Fig. 1 partially broken away to show the interior of the airfoil and, in schematic fashion, the presence of gas bubbles disposed in a blocking material.

Fig. 3 is a view taken along the lines 2-2 of Fig. 1.

Fig. 4 is a cross sectional view taken along the lines 3-3 of Fig. 3 with portions of the airfoil broken away below for clarity.

Fig. 5 is an enlarged view of a portion of Fig. 2.

DETAILED DESCRIPTION OF THE INVENTION

Fig. 1 is a side elevation view of an airfoil for a gas turbine engine as represented by stator vane 10 for such an engine. The term airfoil includes without limitation turbine blades and turbine vanes and refers to all portions of the turbine blade and the stator vane including their flow directing surfaces, platforms and buttresses.

The airfoil has an outer end 12 and an inner end 14. The airfoil has a trailing edge region 16 having a trailing edge 18 which extends in the spanwise direction. A leading edge region 22 is spaced chordwisely forward of the trailing edge region. The leading edge region has a leading edge 24 which extends in the spanwise direction.

The airfoil 10 further has a flow-directing surface 26 which extends between the inner end 14 and the outer end 12. The flow-directing surface includes a suction sidewall 28 and a pressure sidewall 32. The suction sidewall and pressure sidewall are joined at the leading edge 24 and the trailing edge 18 leaving a cavity 34 for cooling air disposed therebetween.

The airfoil 10 has a first internal wall 36 disposed in the cavity 34. The first wall 36 is spaced chordwise from the leading edge 24. The first wall extends between the suction sidewall 28 and the pressure sidewall 32 leaving a first passage 38 for cooling air therebetween. The first passage extends spanwise in the leading edge region 22. During the process of fabrication, cooling air holes 42 are drilled by appropriate means. The cooling air holes extend through the sidewalls to the interior of the airfoil in the leading edge region and in other regions of the airfoil (not shown). The first passage is not in flow communication with the exterior of the airfoil through the leading edge region in a partially formed airfoil.

Fig. 2 is a cross-sectional view of a portion of a partially formed airfoil 10 corresponding to the finished airfoil 10 shown in Fig. 1. A part of the partially formed airfoil is broken away to show the leading edge region of the airfoil. As shown in Fig. 2, the airfoil does not yet have cooling air holes 42 drilled in the leading edge region. The airfoil has a second internal wall 44 extending chordwise to divide the first spanwise passage 38 into a first spanwise region 46 and a second spanwise region 48.

The airfoil 10 has a third internal wall 52 spaced chordwise from the leading edge 24 and from the first wall 36. The third internal wall extends between the sidewalls 28, 32 leaving a second passage 54 extending spanwise adjacent to the first passage 38. A plurality of impingement holes 56 extend

through the first wall to place the first passage in flow communication with the second passage. The ratio of the area of the second passage to the area of a single impingement hole typically exceeds twenty to one (20:1) and may be as high as one-hundred to one (100:1). The size of the impingement hole is about sixty (60) mils high by about twenty (20) mils wide and is larger than the size of the film cooling hole. A first opening 58 at the outer end 12 and a second opening 62 at the inner end 14 of the airfoil place the second passage in flow communication with the exterior of the airfoil. A cover 63 extends over the second opening to seal the opening during manufacture.

A blocking material 64 in the fluid state is shown disposed in the leading edge region 22. Bubbles of gas are trapped during filling of the airfoil with blocking material. The bubbles are shown in schematic fashion as represented by the gas bubble 66, the gas bubble 68 and the gas bubble 72. The first passage 38 is always in fluid communication and generally in gas communication with the second passage 54 through the impingement holes 56 depending on the orientation of the airfoil. As a result, the first passage and the second passage generally have flow communication with the exterior of the airfoil through the first opening 58 at the outer end 12 and the second opening 62 at the inner end 14. However, regions of reduced circulation form for the gas as blocking material in fluid form is flowed during filling into the second passage and into the first passage from the second passage. For example, the location of impingement hole 56a is closest to the second wall, and is spaced spanwise from the second wall. The fluid blocking material might fill to this level leaving a region of restricted circulation R for gas above the hole. The impingement hole places a portion of the second region that is closest to the region R in gas communication with the second passage but does not extend to the region R.

Fig. 3 is a view from above of the first outer end 12 of the airfoil 10 taken along the lines 3-3 of Fig. 1. Fig. 1 and Fig. 3 show the relationship to the outer end 12 of a first dam 74 (shown in phantom on Fig. 1 and in full on Fig. 3) on the suction side of the airfoil. As shown in Fig. 3, a second dam 76 on the pressure side of the airfoil faces the first dam. A cover or a barrier 84 extends over

certain other openings 78, 82 in the airfoil to block the flow of fluid blocking material to these locations. The dams and the barriers cooperate with the first side of the airfoil to form a bowl 86 to receive and funnel the fluid toward the first opening.

5 Fig. 4 is a cross sectional view through the airfoil shown in Fig. 2 along the lines 4-4. The method of drilling a cooling hole 42 at the leading edge 24 of the airfoil employs a laser beam of focused coherent light. The laser beam impacts the exterior of the airfoil.

 Fig. 4 shows the relationship of a laser beam to the exterior of the airfoil and to the first passage 38 during the process of drilling the cooling air hole 42. The laser beam is schematically represented by the center line L. As the laser beam reaches the interior of the airfoil, the laser beam burns through the wall forming a breakout hole on the interior of the wall. The line of strike of the laser beam is coincident with the center line L. The line of the laser beam is shown
10 extending across the first passage 38 to strike the first wall 36 as if blocking material were not present. Alternatively, the laser beam may strike the exterior wall of the airfoil or, depending on the orientation of the cooling air hole, the first internal wall in the leading edge region. Unless blocking material absorbs and dissipates the energy of the beam, Holes may be drilled or partially drilled into
15 the first wall or into the exterior wall in this region. These holes may result in scrapping of the airfoil or costly rework of the airfoil.

 Accordingly, the method of drilling the cooling air hole 42 requires disposing a blocking material in regions of the airfoil adjacent the location of cooling air holes in the leading edge region. This includes orienting the airfoil
20 with respect to a source of blocking material such that the blocking material may be flowed into the second passage 54 of the airfoil and thence into the first passage 38 of the airfoil. Typically, the airfoil is disposed in a fixture for holding the airfoil that tilts the inner end of the airfoil rearwardly about five degrees (5°) from the vertical.

30 In the partially formed condition of the airfoil 10 shown in Fig. 2, the first passage 38 in the leading edge region 22 is a blind cavity having no exit holes

(cooling air holes 42) in the leading edge region. Instead, the impingement holes 56 extending between the first passage and the second passage 54 must provide both an inlet for the blocking material 64 and an exit for trapped gases on the interior of the airfoil.

5 The step of flowing blocking material 64 into the second passage 54 and thence into the first passage 38 typically takes place in a first environment. The term "environment" (such as first environment, second environment, and third environment) is used to refer to the gas which surrounds the exterior of the airfoil during at least a portion of a step and to the pressure and temperature of that
10 gas. The term "gas" includes a gas made up of molecules of a single type of atom, such as argon, or many types of atoms, such as air.

 In one embodiment of the present invention, the gas in the first environment is air at a first pressure P1 and a first temperature T1. The first pressure P1 is atmospheric pressure, approximately fourteen and seven-tenths
15 pounds per square inch absolute (14.7 psia) and the first temperature is approximately equal to room temperature. Room temperature typically lies in a range of about seventy (70) degrees Fahrenheit to eighty (80) degrees Fahrenheit.

 One material which was found acceptable for blocking material is a curable
20 fluid epoxy resin and fluid curing agent. The mixture is cured prior to the drilling operation. Shell Chemical Company, P.O. Box 4320, Houston, Texas, offers this resin as EPON Resin 815 resin. The curing agent is EPI-CURE Curing Agent 3140 material. Equal volume parts of the resin and the curing agent are mixed together. The curing agent is a low viscosity curative, applicable typically to
25 adhesive flooring and general coatings. At temperatures of about eighty (80) degrees Fahrenheit (twenty-five (25) degrees centigrade), the viscosity of the curing agent is about three thousand to six thousand (3,000-6,000) centipoise and the viscosity of the resin is about five hundred to seven hundred (500-700) centipoise. The mixture of the resin and curing agent at room temperature has a
30 viscosity that approximates that of molasses, and may lie in a range of about

seventeen hundred and fifty (1750) centipoise to about thirty-four hundred (3400) centipoise.

Prior to pouring the molasses-like mixture of epoxy resin and curing agent into the first passage, the method includes forming the dams 74, 76 and installing covers at selected locations of the airfoil. For example, the method includes forming the bowl 86 at the outer end of the vane. The bowl is adjacent the first opening 58 of the vane and is formed by applying aluminum tape to the sides of the airfoil. The tape forms a dam 74 on the suction side and dam 76 on the pressure side of the end of the vane. One satisfactory aluminum tape is available as Aluminum Foil Glass Cloth Tape, Part Number 2925-7, from Furon Inc., 14 McCaffrey Street, Hoosick Falls, New York 12090. The height H1 of the tape is about three-eighths ($3/8$) of an inch above the portion of the airfoil which is adjacent to the first opening. The airfoil has other openings 78, 82 at the outer end of the vane. The method includes blocking these other openings if necessary. The same aluminum tape is used for the cover 63 at the second opening 62. The opening is covered prior to flowing blocking material into the second passage 54.

Blocking material is poured into the bowl formed by the dams and into the second passage until the blocking material reaches a height H2 above the opening. The height H2 is less than the height H1 of the dam and typically is about one-fourth ($1/4$) of an inch.

As the blocking material flows into the second passage 54 and into the first passage 38, the blocking material traps gas, such as the air of the first environment, in pockets in the second passage. Alternatively, the aluminum tape might be applied to provide a barrier at the second opening after the blocking material was flowed into the second passage. This would ensure that as much air as possible was driven out of the second passage but is time consuming because the airfoil must be manipulated and cleaned of any resin that reaches the exterior of the airfoil. This would occur prior to substantial amounts of the blocking material flowing into the first passage.

Enough fluid blocking material is mixed so that it is all substantially consumed in filling the next batch of airfoils going into the vacuum oven. Thus, the amount of time filling a batch of the airfoils is about equal to the time spent in the partial vacuum by the previous batch of filled airfoils.

5 The airfoil filled with blocking material is disposed in a vacuum oven having a second environment of air at a second temperature T2 and a second pressure P2 to form a partial vacuum for a period of about twenty (20) to thirty (30) minutes. The second pressure P2 is less than the first pressure P1 (typically, atmospheric pressure) of the first environment. The second pressure
10 is typically less than one hundred and twenty-five (125) Torr that is, less than about two and one half (2.5 psia). The temperature T2 of the atmosphere at the partial vacuum is about room temperature. The temperature T2 in alternate embodiments of the invention might increase above the temperature T1 as long as hardening of the blocking material is not increased by an amount that would
15 capture air bubbles in the leading edge region as the air is evacuated from the blocking material into the second environment. A temperature T2 higher than the temperature T1 might be advantageous in situations where the viscosity of the blocking material decreases with temperature.

 The method includes orienting the airfoil such that buoyant forces act on
20 the bubbles of gas trapped in the blocking material. In the embodiment shown, the vane is a free standing structure supported by its inner end and may be placed in the vacuum oven without further support. Alternatively, the vane is disposed in a fixture which cants the second end of the vane rearwardly about five (5) degrees. Once such fixture has a pair of parallel rods which each
25 engage the underside of the outer end of the vane platform. This avoids the need for special fixturing, speeding handling of the vanes and reducing handling costs.

 The partial vacuum causes the air bubble 68 to expand, increasing the buoyant forces acting on the bubble. An expanded bubble in the first passage
30 38 is urged by buoyant forces from the first passage into the second passage 54. This expanded bubble joins other expanded bubbles disposed in the second

passage. As the expanded bubbles, such as the bubble 73, flow upwardly in the second passage, the bubbles urge the blocking material to flow into the first passage through the impingement holes 56 extending between the second passage and the first passage. Blocking material urged into the first passage in turn helps force air from the first passage through the impingement holes into the second passage increasing the effect of the bubbles passing through the second passage and urging more blocking medium into the first passage.

After disposing the airfoil filled with blocking material in the partial vacuum of the second environment, the airfoil filled with blocking material is disposed in a third environment. The third environment has a gas at a third pressure P_3 which is greater than the second pressure P_2 . The gas is typically air at a temperature T_3 which is greater than the temperature T_2 . For example, the temperature used with this particular mixture of epoxy resin and curing agent (EPON Resin 815 resin and EPI-CURE Curing Agent 3140 material) is in a range of about two hundred and twenty-five (225) degrees Fahrenheit to two hundred and seventy-five (275) degrees Fahrenheit. Typically the environment is at a temperature of about two hundred and fifty (250) degrees Fahrenheit. The mixture of epoxy resin and curing agent is held in this temperature range for about twenty (20) minutes to thirty (30) minutes causing the mixture to harden. This is followed by a curing time in a fourth environment of about forty-eight hours at room temperature and atmospheric pressure to finally cure the mixture of epoxy resin and curing agent. An alternate embodiment of the present invention might use another material such as wax. In such a case, the step of flowing the blocking material into the second passage in a first environment might be carried out at a higher temperature. Another approach would use wax in the molten state at such an elevated temperature that the heat in the wax keeps the wax molten as it is flowed into the vane. The step of disposing the filled airfoil in a second environment would be carried out in a vacuum oven in a partial vacuum at an elevated temperature to keep the wax molten. The step of disposing the airfoil filled with wax in the third environment for curing would take

place at a temperature T_3 lower than the temperature T_2 and at a pressure P_3 which is higher than the pressure P_2 to solidify the wax.

A particular advantage of the present method occurs in filling articles with blocking material that have regions of reduced circulation such as the region R shown in Fig. 2 and shown in an enlarged view in Fig. 5. Although the phenomena is not well understood, empirical data suggests that the following occurs in this region. It is possible for the region R to trap a bubble of gas which is coextensive in spanwise height with the height B of the region R. Disposing the airfoil filled with blocking material in the second environment at a pressure which is less than approximately one hundred and twenty-five (125) Torr, decreases the external pressure of the gas in the environment on the blocking material. This, in turn, decreases the pressure on the trapped gas in the bubble.

The ideal-gas law provides a good approximation of the effect of pressure of the environment and temperature of the blocking material on the volume of the trapped bubble.

The ideal-gas law states that $pV = nRT$ where V = volume of n moles of gas; p = absolute pressure; T = absolute temperature; and R = a universal constant whose value depends only on the units used. Gases at a pressure of a few atmosphere may usually be considered ideal for engineering calculations. The ideal-gas law suggests that decreasing the external pressure by a factor of six (from seven hundred and sixty (760) Torr to less than one hundred and twenty-five (125) Torr) will result in a sixfold increase in volume of the trapped gas. In the absence of the array of impingement holes 56, the bubble would expand spanwisely to a spanwise height B_2 in the second portion of the passage. The new spanwise height B_2 with the partial vacuum would be about six (6) times greater than the spanwise height B_1 .

Impingement holes are not absent from the airfoil. In fact, there are impingement cooling air holes which place the first passage in flow communication with the exterior of the airfoil. Decreasing the pressure to pressure P_2 allows the trapped gas to expand to the impingement hole 56a which is closest to the second wall 44 and which is adjacent to the region R. The

buoyant forces acting on the trapped bubble 72 push the bubble out through the impingement hole 56a and into the second passage and thence to the exterior of the airfoil. This continues until the gas bubble no longer extends out of the region R. Thus at the reduced pressure P_2 , the trapped gas bubble still extends
 5 for the same spanwise height B_1 but many molecules of air have escaped.

Disposing the airfoil filled with a blocking material in the third environment at the third pressure P_3 causes the volume of the gas to decrease to a spanwise height B_3 and to less than a sixth of its original volume. Thus, the trapped gas still displaces a small portion of the blocking material and forms a void of height
 10 B_3 . However, the shrunken void is not aligned with the drilling axis L of the laser beam for the cooling air hole 42a which extends through the leading edge of the airfoil between the impingement hole and the second wall 44. Heating the blocking material during the hardening step does cause the volume of the gas to increase above the temperature T_2 . The volume increase is small
 15 (approximately a two to threefold increase in volume depending on the temperature of the gas in the bubble) compared to the decrease in volume that results from decreasing the pressure to a sixth ($1/6$) of its normal value during the process of removing the trapped gas. As a result, the laser beam L does not intercept the void formed by the bubble and the laser beam does not strike the
 20 wall.

This method may also be applied to any blocking material which is fluid in form and hardened. For example, a blocking material which has a wax base is evacuated in the second environment at an elevated temperature to keep the wax in fluid form and then cooled to room temperature to harden the wax. For
 25 such a blocking material, both the pressure and temperature of the second and third environments act to reduce the volume of the trapped gas during the evacuation and hardening process.

The hardening process is completed by curing the resin mixture at room temperature and atmosphere pressure for forty-eight (48) hours in a fourth
 30 environment. Once the hardening process is complete, the blocking material maintains its shape by reason of its increased resistance to differences in

ambient pressure. The epoxy resin is particularly resistant to any effect of pressure but it is believed that wax-like material would work equally well.

The method includes the step of drilling cooling air holes through the external wall of the airfoil into the leading edge cavity and at such other locations
5 as may be required. This step is carried out using conventional laser drilling techniques.

Thereafter, the interior of the airfoil is cleaned of the blocking material by placing the airfoil in a fifth environment having air, such as an oven, and heating the environment to a temperature of thirteen hundred (1,300) degrees
10 Fahrenheit. The epoxy resin is vaporized leaving at most some fine particles which may pass through the cooling air holes. This is followed by performing a water flow test on the airfoil to ensure that the cooling air holes are of acceptable size. The water flow test flows water through the airfoil and would carry any oxidized particles of resin away if any such particles exist.

15 An advantage of the present invention is the reduced cost of fabricating the airfoil which results from using a blocking material which does not require extensive manipulation of the airfoil in many directions to remove trapped gas bubbles with buoyant forces. Another advantage in one embodiment is the reduced level of risk of scalding personnel handling the airfoil during the filling
20 operation which results from using an epoxy resin and curing agent at room temperature. Another advantage is the level of cost for the process which results from avoiding costs associated with scrapped airfoils or the costs associated with reworking airfoils which may be damaged during the drilling operation but which are reworked to bring them into acceptable configuration.
25 Still another advantage is the removal of the blocking material by heating the airfoil to an elevated temperature which vaporizes most, if not all, of the blocking material. Any fine particles of powder left behind are easily washed away during further processing of the airfoil.

Although this invention has been shown and described with respect to the
30 detail embodiments thereof, it will be understood by those skilled in the art that

various changes in the form and detail thereof may be made without departing from the spirit and scope of the claimed invention.

What is claimed is:

1. A method for disposing a fluid, hardenable material in a structure having a shaped cavity prior to laser drilling through structure adjacent the cavity, the structure having an internal configuration which causes the fluid material to trap a pocket of gas and having an opening for allowing gas to escape from the cavity, comprising:
- 5 flowing fluid, hardenable material into the cavity in a first environment having a gas at a first pressure P_1 and a first temperature T_1 , the fluid, hardenable material being disposed in the cavity such that a pocket of gas is partially bounded in the cavity by the hardenable material;
- 10 disposing the structure filled with the fluid, hardenable material in a second environment having a gas at a second temperature T_2 that does not cause the material to harden to an extent that traps bubbles during the step and a second pressure P_2 that is less than the first pressure P_1 , and expanding the gas in the pocket causing the volume of the gas in said
- 15 pocket to become larger than the pocket and forcing part of the trapped gas out of the pocket and into gas communication with the opening, including
- orienting the structure such that buoyant forces acting on said gas in the cavity urges said gas toward said first opening;
- 20 disposing thereafter the structure filled with fluid, hardenable material in a third environment at a third pressure P_3 which is greater than the second pressure P_2 for a predetermined period of time to harden the blocking material;
- wherein any pocket of gas in the cavity will expand from the pressure P_1 against the lower pressure P_2 and at least a portion will escape from the cavity through the opening; and, wherein the higher pressure P_3 acting on the blocking material compresses any bubbles remaining to reduce the size of a void in the hardened blocking material in the cavity such that the void in the cavity forms out of the line of the laser beam in the cavity to such an extent that during laser drilling the
- 30 blocking material intercepts the laser beam to prevent unacceptable damage to the structure.

2. The method for disposing a fluid, hardenable material in a structure having a shaped cavity of claim 1 wherein the gas in the first, second and third environments is air.
- 5 3. The method for disposing a fluid, hardenable material in a structure having a shaped cavity of claim 2 wherein the pressure P_2 is a partial vacuum.
4. The method for disposing a fluid, hardenable material in a structure having a shaped cavity of claim 1 wherein the pressure P_2 is a partial vacuum.
- 10 5. The method for disposing a fluid, hardenable material in a structure having a shaped cavity of claim 3 wherein the pressure P_2 is a partial vacuum at a pressure less than about one-hundred and twenty-five Torr.
6. The method for disposing a fluid, hardenable material in a structure having a
15 shaped cavity of claim 4 wherein the pressure P_2 is a partial vacuum at a pressure less than about one-hundred and twenty-five Torr.
7. The method for disposing a fluid, hardenable material in a structure having a shaped cavity of claim 1 wherein the second temperature T_2 of the gas in the
20 second environment is at a temperature which is greater than the temperature T_1 .
8. The method for disposing a fluid, hardenable material in a structure having a shaped cavity of claim 7 wherein the blocking material is a wax based material having a melting point above the first temperature T_1 and is flowed in molten form
25 into the structure.
9. The method for disposing a fluid, hardenable material in a structure having a shaped cavity of claim 1 wherein the second temperature T_2 of the gas in the

second environment is at a temperature which is approximately the same as the temperature T_1 .

10. The method for disposing a fluid, hardenable material in a structure having a
5 shaped cavity of claim 9 wherein the second temperature T_2 of the gas in the
second environment and the temperature T_1 of the gas in the first environment
are each about room temperature.

11. The method for disposing a fluid, hardenable material in a structure having a
shaped cavity of claim 10 wherein the blocking material is an epoxy resin material
10 which is mixed at room temperature and thereafter flowed into the airfoil without
further heating.

12. The method for disposing a fluid, hardenable material in a structure having a
shaped cavity of claim 1 wherein the third pressure P_3 of the gas in the third
15 environment is at a pressure which is approximately the same as the pressure P_1
of the first environment.

13. The method for disposing a fluid, hardenable material in a structure having a
shaped cavity of claim 12 wherein the third pressure P_3 of the gas in the third
20 environment and the pressure P_1 of the gas of the first environment are ambient
pressure.

14. The method for disposing a fluid, hardenable material in a structure having a
shaped cavity of claim 1 wherein the third temperature T_3 of the gas of the third
25 environment is at a temperature which is greater than the temperature T_2
and provides a hardening temperature for the blocking material.

15. The method for disposing a fluid, hardenable material in a structure having a
shaped cavity of claim 14 wherein the material is an epoxy resin and the

temperature is the setting temperature for the resin of about two-hundred and fifty (250) degrees Fahrenheit.

16. The method for disposing a fluid, hardenable material in a structure having a shaped cavity of claim 1 wherein the material is flowed in molten form into the structure and wherein the third temperature T_3 of the gas of the third environment is at a temperature which is less than the melting temperature of the material to solidify the material.
17. The method for disposing a fluid, hardenable material in a structure having a shaped cavity of claim 1 wherein the opening is a first opening and a vent hole is adjacent to the first cavity and is in gas communication with the first opening and an expanded bubble from the cavity and wherein the step of forcing part of the gas out of the pocket includes flowing that part of the gas through the vent hole to enable at least a portion of the bubble to escape from the first cavity through the vent hole.